A Space Mission to Test MOND and the Pioneer Anomaly

Abstract.

In light of current efforts to understand the Pioneer Anomaly (Anderson et al., 1998, 2002a, 2002b, 2002e) we offer a testable explanation involving mModified Newtonian dDynamics (MOND)(Milgrom 1994). We are suggest that radial trajectories, in otherwise unmodified gravitational potentials, introduce as modified inertia responsible for dynamic anomalies dynamical. As eExamples of radially evolving systems, we examine the Pioneer Anomaly, and the cosmological acceleration observed using Type Ia supernovae (Riess et al., 1998; Perlmutter et al. 1999a, 1999b). We find that MOND predicts observable effects, which current laboratory searches forstudies of modified inertia (i.e., the sStrong eEquivalence pPrinciple) has not been sensitive todo not detect. We describe how the addition of a second space_probe to the proposed Anderson et al. Pioneer Anomaly mission proposed by Anderson et al. would constrain the prevailing MOND models.

Introduction

Anderson et al. (1998, 2002a) finds have-found that the Pioneer 10, Pioneer 11, Galileo, and Ulysses deep space probes shared an anomalous, constant acceleration of magnitude $|\overline{\alpha_{\vec{P}}}| = (8.74 \pm 1.33) \times 10^8$ cm/s/s_s directed radially towards the Sun. This "—Pioneer Anomaly" is apparently not because ofdue to mission systematics (Anderson et al., 2002b; Murphy et al., 1999; Katz et al., 1999; Anderson et al., 1999a, 1999b) and may require new physics in order to be accurately modelled. -We consider an intriguing explanation involving new physics: the — mModified Newtonian dDynamics (MOND) theory, due todeveloped by Milgrom (1983, 1986, 1989; Bekenstein & Milgrom, 1984).

We_will employ the MOND formalisms of modified inertia so that the -dynamical bodies move in trajectory-dependent effective potentials. MOND is parameterized by a characteristic acceleration $a_0 \sim 10^{-8}$ cm/s, which is usually small compared to the expected Newtonian acceleration a_N .

<u>The mMagnitude of the MOND contributions too observable dynamics are is given by the</u> heuristic function $\mu(a/a_0)_{\pm}$ for which where $\mu \approx 1$ when $a/a_0 \gg 1$ and $\mu \approx a/a_0$ when $a/a_0 \leq 1$. In this paper, we seek, to discover a dynamical dependence in: the vector *direction* of a trajectory; we suggest an explanation for the Pioneer Anomaly involving MOND effects on *radial* trajectories **Commented [CP1]:** The style guide of your target journal, JCP, does not allow citations in the abstract, so I have removed them.

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(i.e., $v \cdot a \approx 1$).

Already Anderson et al. (2002c) <u>have previouslys</u> described a deep space probe mission capable for sensitively figuring out,of detecting the magnitude of the Pioneer Anomaly with high <u>sensitivity</u>. However, since the cause, of the <u>a</u>Anomaly is unknown, <u>this</u> mission <u>has ais</u> purely empirical <u>in</u> nature. Here, we describe (1) an explanation for the Pioneer Anomaly <u>involving using</u> MOND and (2) an easy addition <u>of to</u> the Anderson et al. mission to test our MOND hypothesis.

The Equivalence Principle

The sector get quivalence pPrinciple (SEP) says that the gravitational mass, m_{gs} and the inertial mass, m_{is} of a body are — identical. Experiments to verify the SEP typically quantify the $\eta \equiv \Delta a/a$. Experiments to verify the SEP typically quantify the $\eta \equiv \Delta a/a$. Experiments to verify the SEP typically quantify the $\eta \equiv \Delta a/a$. Between the attractive and dynamical accelerations. Su et al. (1994) and Smith et al. (2000) have observed Laboratorylaboratory-scale masses have been observed (Su et al., 1994; Smith et al., 2000) to havo fe $\eta < 10^{-13}$. The iInteractions about of self-gravitating bodies determine are a special problem (Anderson et al., 1996), and having a it's the current observational limit is of $\eta < 10^{-6}$ (Milani et al., 2002), although the Earth — Moon system has an $\eta < 10^{-13}$ (Anderson & Williams, 2001).

Current experimental designs to that aretesttest the SEP do not consider trajectory dependence. By performing aA literature review revealed, we found that experiments and, if facted the and found examine involved predominantly involved azimuthal trajectories. For example, the experiments of Su et al. (1994) SEP experiments test-ed horizontal accelerations in a terrestrial laboratory₂:- the The sensitivitye limit of η wais measured for accelerations towards the Sun using masses orbiting with the Earth on azimuthal trajectories. Measurements of the SEP in of the <u>a</u>-selfgravitating objects, like-such as a-planets (Anderson et al., 1998, 1996; Milani et al., 2002; Anderson & Williams, 2001) and the neutron stars (Darmour & Schaefer, 1991), is are also limited to azimuthal orbital trajectories azimuthal orbital.

Radial_-trajectory experiments (Kuroda & Mio, 1989; Dittus & Mehls, 2001; Reasenberg & Phillips, 2001)₅ present the <u>a</u> greater experimental challenge (Blaser, 2001)<u>.</u> that has <u>These</u> <u>measurements have</u> yielded <u>a</u> less sensitive upper limit_a $\eta \le 10^{-9}$. <u>They eBased on these results</u>, we would not expects that these laboratory experiments could detect that the Pioneer

Commented [CP3]: When a theory, model, or procedure is named after two or more individuals, their names are conventionally joined using an en dash. The same rule applies whenever the elements of a compound are considered equivalent, so parent-child relationship, cost-benefit analysis, etc., use an en dash rather than a hyphen.

Commented [CP4]: Edits to this sentence may have changed your intended meaning; please review before accepting changes. Anomaly would to be detectable in this these laboratory experiments.

MOND

In-To trying to modify the inertia, we don't want to alter the Newtonian kinetic-action energy

$$S_N = \frac{1}{2}m \int v^2 dt$$

such that <u>a</u> radial trajectory dependence is introduced while Newtonian dynamics are still recovered in the limit $a_0 \rightarrow 0$.

This is most simply accomplished most simply with an action equation of the form:

$$S_M = \frac{1}{2}m \int \left[1 - \underbrace{\mu \; \frac{a_0}{a} \; (\hat{v} \cdot \hat{a})}_{M_S}\right] v^2 dt$$

The additional factor, M_{S_3} vanishes for any near-circular orbit of azimuthal trajectory $v \cdot a \approx 0$, which such as those of the, planets. Indeed, Anderson et al. (1998) are have calculated that any universally eaffective property of the gravitational force; capable of producing the Pioneer deceleration would already be have been sobserved observed in the orbital motions of the planets. (AlsoIn addition, this form of M_S doesn't does not produces effects for on the orbits of stars in galaxies, and of or for galaxies in clusters, as was the original intention of MOND. These effects This could be included in a more complicated form of M_{S_3} but we consider these "''dark matter''" issues (Castillo-Morales & Schindler, 2003) to be a separate problem not of interest hereoutside the scope of this work.)

Our modified action equation predicts MOND effects for all {\em all} radial trajectories.

For the large accelerations- $a/a_0 \gg 1$ $a/a_0 \gg 1$, $\mu \approx 1$, and $M_S \propto a_0/a$. For the Pioneer Anomaly, we expect $M_S \sim 10^{-4}$, as has been observed. Furthermore, the anomalous Pioneer acceleration, a_{P_2} is *constant*, meaning $\eta_P \propto r^2$, as expected from for thean M_S for with a constant a_0 and $a = GM/r^2$. For small accelerations, $a/a_0 \leq 1$, $\mu \approx a/a_0$, and $M_S \approx 1$. You This effect can also use this to explain the anomalously faint, high-redshift Type Ia supernovae (SNIa) **Commented [CP5]:** Edits to this sentence may have changed your intended meaning; please review before accepting changes.

Commented [CP6]: Please consider whether "sophisticated" might be a better word choice here. observed by Riess et al. (1998) and Perlmutter et al. (1999a, 1999b) is due toas small-acceleration MOND effects on the radial trajectories of the cosmological expansion.

Table 1. Accelerations				
Body	Potential	η	a_N	a_0/a
			$({\rm cm} {\rm ~s}^{-2})$	
azimuthal systems with $\hat{v} \cdot \hat{a} \approx 0$				
Moon	Earth	$< 10^{-13}$	0.3	$3.3 imes 10^{-8}$
SEP lab	Sun	$< 10^{-13}$	0.6	$1.7 imes 10^{-8}$
Mars	Sun	$< 10^{-6}$	0.3	$3.3 imes 10^{-8}$
star	galaxy	~ 1	2×10^{-8}	0.5
galaxy	cluster	~ 1	2×10^{-10}	50
radial systems with $ \hat{v} \cdot \hat{a} \approx 1$				
Pioneer	Sun (at 30 AU)	-10^{-4}	7×10^{-4}	2×10^{-5}
SEP lab	Earth	$< 10^{-9}$	$9.8 imes 10^2$	10^{-11}
cluster	Universe	~ 1	7×10^{-8}	0.2

the Table 1 summarizes the MOND regime of various known experimentally determined accelerations. Of the interest here are the The final three entries rows, which comprise our knowledge of the radial trajectories.

The rightmost column lists the value of a_0/a_{\pm} which is the detection threshold of M_S in the strong acceleration limit. Anderson et al. say have suggested that detecting the Pioneer Anomaly requires acceleration measurements accurate to at least one part in 10⁶, consistent with our predicted MOND contribution. Furthermore, we may predict that radial_trajectory SEP experiments in terrestrial laboratories will detect MOND effects when the accuracyies reaches one part in 10¹² (which is still three orders of magnitude away). However, the The manifestation of MOND manifestation in the Pioneer Anomaly, on the other hand, is readily testable with current technology.

The Experiment

Anderson et al. (2002c) haves proposed the an experiment to further characterize the Pioneer Anomaly. Since the proposed mission uses a radial trajectory, we offer suggest a modification to test our MOND hypothesis: <u>uUse a twin spacecraftspacecrafts</u> in the <u>a</u> near-circular orbit. Placing a twinone space probe on an azimuthal trajectory, should simply demonstrate the radial dependence of the MOND interpretation of the Pioneer and SNIa anomalies. As originally planned, tThe Anderson et al. radial mission_-would sensitively probe the known anomaly with high sensitivity, while our azimuthal mission should return a null result. The azimuthal mission_, like the <u>a</u> planets, will have a near-circular trajectory, like that of a planet, that is presumably unaffected by the **Commented [CP7]:** Please confirm that this number is consistent with Table 1.

proposed inertial MOND contributions. Since the Pioneer Anomaly is best observed beyond ~20 AU, we suggest that the two missions share radial trajectories out to the orbit of Neptune $(30 \text{ AU})_{a}$ at which point the azimuthal mission can be gravitationally deflected by Neptune into a bound, low-eccentricity orbit (Fig. 1). The deflected orbit may be chosen to be out<u>side</u> of the ecliptic plane (to characterize external heliospheric accelerations, to-probe the solar system potential at various angles, or simply-to study the heliosphere at unprecedented radii and latitudes).



Figure 1. Suitable mission trajectories for (a) the Anderson et al. probe and (b) the azimuthal twin. The planetary positions are correct for June 2003.

The shared radial portions, of <u>the</u> missions should provide <u>reliable accounts</u> a reliable accounts of (1) the measurements of the anomaly and (2) the similarity of the <u>two</u> spacecrafts. After a deflectionsng from Neptune, the predicted disappearance of the anomaly in the azimuthal probe would <u>provides provide</u> significant evidences against the interpretation of the anomaly as an onboard systematic <u>effect</u>. of the twin probes.

The trajectory dependence in the probe's' accelerations of the probes would be easily detectable using standard telemetry. In addition, the large proper motion of the azimuthal mission would be detectable using very long baseline radio interferometric techniques, allowing <u>for</u> independent verification of the telemetry calculations (Anderson et al. 2002b).

Conclusion

We find that <u>the</u> radial_-trajectory <u>phenomenasphenomena</u> are subject to deviations from Newtonian dynamics due to MOND_-modified inertia. The term M_{S^-} which modifies the kinetic action, possesses a simple algebraic form which that follows naturally from the trajectory constraints and existing experimental limits.

Currently, this <u>anomaly</u> is best suited <u>for to observation in the <u>a</u> space-borne experiments. We propose <u>testing</u> <u>s</u> to test, MOND effects in the vicinity of our Sun using <u>the</u> space flight</u> described by Anderson et al. with the addition of a twin probe deflected into <u>a</u> closed orbit at Neptune.